Contents lists available at ScienceDirect

# Journal of Computational Physics

journal homepage: www.elsevier.com/locate/jcp



# Enablers for robust POD models

## M. Bergmann<sup>\*</sup>, C.-H. Bruneau, A. Iollo

INRIA Bordeaux Sud Ouest, Team MC2 and Institut de Mathématiques de Bordeaux, Université Bordeaux 1, UMR 5251 CNRS, 351, Cours de la Libération, 33405 Talence cedex, France

#### ARTICLE INFO

Article history: Received 13 June 2008 Received in revised form 17 September 2008 Accepted 29 September 2008 Available online 11 October 2008

Keywords: Proper orthogonal decomposition Reduced order model Stabilization Functional subspace improvement

#### ABSTRACT

This paper focuses on improving the stability as well as the approximation properties of reduced order models (ROMs) based on proper orthogonal decomposition (POD). The ROM is obtained by seeking a solution belonging to the POD subspace and that at the same time minimizes the Navier–Stokes residuals. We propose a modified ROM that directly incorporates the pressure term in the model. The ROM is then stabilized making use of a method based on the fine scale equations. An improvement of the POD solution subspace is performed, thanks to a hybrid method that couples direct numerical simulations and reduced order model simulations. The methods proposed are tested on the two-dimensional confined square cylinder wake flow in laminar regime.

© 2008 Elsevier Inc. All rights reserved.

#### 1. Introduction

#### 1.1. Reduced order models based on proper orthogonal decomposition

In the last decades, the conception and the optimization of the aerodynamics/aeroacoustics of ground vehicles and airplanes have been pursued by numerical simulation. The applications mainly concern unsteady turbulent flows that develop at high Reynolds numbers. The numerical simulation of such flows, as well as their control, requires massive computational resources. Indeed, after discretization of the governing equations, *i.e.* the Navier–Stokes equations in fluid mechanics context, one must then solve a system of equations whose complexity algebraically grows with the number of degrees of freedom of the system to be solved. Now, and despite the considerable progress made in the numerical field (power of the computers, new and more efficient algorithms), it is still very difficult to solve such large problems for complex flows in real time, that is, *in fine*, a major stake for industrials. To overcome this difficulty, it is possible to determine a reduced order model of the flow dynamics keeping only few adapted modes. The choice of these modes is not unique, and it strongly depends on the characteristics of the flow that one wants to approximate, or even might depend on some expected outputs (goal-oriented models [1]). Several methods are commonly used, among them are proper orthogonal decomposition (POD) [2–4], balanced truncation [5–7], global eigenmodes [8], Galerkin modes [9], *etc.* Due to the energetic optimality of its basis, the POD is chosen in this study. By this technique, it is possible to extract the dominant characteristics (POD modes) of a given database, and the ROM is then obtained thanks to a Galerkin projection of the governing equations onto these modes. Although this method for reducing the order of a system can be very efficient in some flow configurations, it also presents several

\* Corresponding author. *E-mail address:* Michel.Bergmann@inria.fr (M. Bergmann).

<sup>0021-9991/\$ -</sup> see front matter @ 2008 Elsevier Inc. All rights reserved. doi:10.1016/j.jcp.2008.09.024



Fig. 1. Flow configuration and vorticity snapshot at *Re* = 200.

drawbacks. Besides the possible inherent lack of numerical stability of POD/Galerkin methods, even for simple systems [10], the main shortcomings are the following:

- Since in most of the POD applications for incompressible flows the POD ROM is built from a velocity database,<sup>1</sup> it is necessary to model the pressure term. Usually, in many closed flows, the contribution of the pressure term formally drops out due to fortunate choices of boundary conditions in the POD ROM. However, for convectively unstable shear layers, as the mixing layer or the wake flow, it was proved in [11] that neglecting the pressure term may lead to large amplitude errors in the Galerkin model. Therefore, to accurately model such flows, the pressure term [11,12] must be modeled. To overcome this difficulty, a pressure extended reduced order model is introduced in Section 3, so that the pressure term can be directly approximated using the pressure POD mode.
- Due to the energetic optimality of the POD basis functions, few modes are sufficient to give a good representation of the kinetic energy of the flow.<sup>2</sup> For model reduction purpose, we only keep these few modes that are associated to the large eddies of the flow (as the vortices of the Von Kármán street that usually develop behind bluff bodies). But since the main amount of viscous dissipation takes place in the small eddies represented by basis functions that are not taken into account, the leading ROM is not able to dissipate enough energy. It is then necessary to close the ROM by modeling the interaction between the calculated modes and the non-resolved modes. This problem is similar to that of large eddy simulation (LES) [13] of turbulent flows. In this study, the ROM is closed using Navier–Stokes equations residuals and exploiting ideas similar to streamline upwind Petrov–Galerkin (SUPG) and variational multiscale (VMS) methods [14].
- Since POD basis functions are optimal to represent the main characteristics included in the snapshot database of the flow configuration used to build them, the same basis functions are *a priori* not optimal to efficiently represent the main characteristics of other flow configurations. Indeed, for flow control purpose, it was demonstrated [15–17] that POD basis functions built from a flow database generated with a given set of control parameters is not able to represent the main features of a flow generated with another set of control parameters. To overcome this problem, we propose to derive methods allowing to adapt the POD basis functions at low numerical costs. This is the central question of Section 5.

#### 1.2. Flow configuration

In this study, the confined square cylinder wake flow (Fig. 1) is chosen as a prototype of separated flow. This flow is interesting since it presents detachments of the boundary layer, wake and vortices interactions with walls. The Navier–Stokes equations, written in their dimensionless and conservative form, are written as

$$\frac{\partial \boldsymbol{u}}{\partial t} + (\boldsymbol{u} \cdot \boldsymbol{\nabla})\boldsymbol{u} = -\boldsymbol{\nabla}\boldsymbol{p} + \frac{1}{Re}\Delta\boldsymbol{u}, \tag{1a}$$
$$\boldsymbol{\nabla} \cdot \boldsymbol{u} = \boldsymbol{0}, \tag{1b}$$

where  $Re = U_{\infty}L/v$  denotes the Reynolds number, with  $U_{\infty} = u(0, H/2)$  the maximal inflow velocity, *L* the length of the side of the square cylinder and *v* the kinematic viscosity. In what follows, we consider Re = 100 and Re = 200, that is to say, the laminar regime. Otherwise, the same parameters as those introduced in [12] are used in this study, *i.e.* the blockage ratio  $\beta = L/H$  is equal to 0.125 and the domain  $\Omega$  is  $(0, 4H) \times (0, H)$ . The same numerical method as that described in [12] is used. A vorticity representation of a flow snapshot is presented in Fig. 1 (dashed lines represent negative values). The boundary layer detachment, the wake and the vortex interactions with top and bottom walls are visible.

This paper is organized as follows. Section 2 presents the proper orthogonal decomposition (POD, Section 2.1) and the standard velocity POD/Galerkin reduced order model (POD ROM, Section 2.2) for incompressible flows. A pressure extended reduced order model is introduced in Section 3. Different stabilization methods of the POD ROM are presented in Section 4. A residuals-based stabilization method (Section 4.1), and streamline upwind Petrov–Galerkin (SUPG) as well as the variational

<sup>&</sup>lt;sup>1</sup> In almost all the experimental works, the pressure field is unavailable.

<sup>&</sup>lt;sup>2</sup> This is true for 2-D periodic laminar flows, but thousands of POD modes could be necessary to describe the fluctuation energy of a fully developed turbulent flow.

multiscale (VMS) methods (Section 4.2) are introduced. Section 5 presents methods to adapt the functional subspace when input system parameters change. A Krylov-like method (Section 5.1) and a hybrid DNS/POD ROM method (Section 5.2) are presented. Finally, Section 6 is dedicated to conclusions.

#### 2. Standard reduced order model based on proper orthogonal decomposition

#### 2.1. Proper orthogonal decomposition

The proper orthogonal decomposition (POD) was first introduced in turbulence by Lumley [18] in 1967 as an unbiased definition of the coherent structures widely known to exist in a turbulent flow. A comprehensive review of the POD can be found in Refs. [2–4]. The POD, also known as Karhunen–Loéve decomposition, principal component analysis or empirical eigenfunctions method, consists of looking for the deterministic function  $\Phi(\mathbf{x})$  that is most similar in an average sense to the realizations  $U(\mathbf{x}, t)$ . For instance, the realizations  $U(\mathbf{x}, t)$  can be velocity fields, pressure fields, temperature fields, etc. Since in this study the data are issued from numerical simulations, the method to compute POD modes introduced by Sirovich [3] is adopted (see [4] for justifications). In this case, the constrained optimization problem reduces to the following Fredholm integral eigenvalue problem

$$\int_0^1 C(t,t')a_n(t')dt' = \lambda_n a_n(t),$$
(2)

where the temporal correlation tensor  $C(t, t\nu)$  is defined by

$$C(t,t') = \frac{1}{T} (\boldsymbol{U}(\boldsymbol{x},t), \boldsymbol{U}(\boldsymbol{x},t'))_{\Omega}.$$
(3)

The inner product  $(.,.)_{o}$  between two fields (*U* and *V* is computed as

$$(\boldsymbol{U}, \boldsymbol{V})_{\Omega} = \int_{\Omega} \boldsymbol{U} \cdot \boldsymbol{V} \mathrm{d}\boldsymbol{x} = \int_{\Omega} \sum_{i=1}^{n_c} U^i V^i \mathrm{d}\boldsymbol{x},$$

where  $U^i$  represents the *i*th component of the vector **U** with dimension  $n_c$ .

The eigenvalues  $\lambda_n$  (n = 1, 2, ...) determined in (2) are all real and positive, and form a decreasing and convergent series. Each eigenvalue represents the contribution of the corresponding mode  $\Phi_n$  to the information content of the original data. Note that if **U** are the velocity fields, the information content reduces to the kinetic energy.

In Eq. (2),  $a_n$  are the time-dependent POD eigenfunctions of order *n*. These modes form an orthogonal set, satisfying the condition

$$\frac{1}{T}\int_0^1 a_n(t)a_m(t)\mathrm{d}t = \lambda_n \delta_{nm}.$$
(4)

The associated eigenvectors  $\Phi_n$  (also called empirical eigenfunctions) form a complete orthogonal set and are normalized, so that they verify  $(\Phi_n, \Phi_m)_{\Omega} = \delta_{nm}$ .

The spatial basis functions  $\Phi_n^i$  can then be calculated from the realizations  $U^i$  and the coefficients  $a_n$  with

$$\Phi_n^i(\boldsymbol{x}) = \frac{1}{T\lambda_n} \int_0^T U^i(\boldsymbol{x}, t) a_n(t) \mathrm{d}t.$$
(5)

Since the POD eigenfunctions can be represented as linear combinations of the realizations, they inherit all the properties of the original data. For instance, the eigenfunctions are divergence free for incompressible flows. Moreover, the eigenfunctions verify the boundary conditions of the numerical simulation used to determine the flow realizations.

The set of POD modes  $\{\Phi_n\}_{n=1}^{N_{POD}}$  is complete in the sense that any realization  $\boldsymbol{U}(\boldsymbol{x},t)$  contained in the original data set, can be expanded with arbitrary accuracy (in function of  $N_{POD} \ge 1$ ) in the eigenfunctions as

$$\boldsymbol{U}(\boldsymbol{x},t) \simeq \widehat{\boldsymbol{U}}^{[1,\dots,N_{POD}]}(\boldsymbol{x},t) = \sum_{n=1}^{N_{POD}} a_n(t) \boldsymbol{\Phi}_n(\boldsymbol{x}).$$
(6)

For later convenience, the estimation  $\widehat{U}^{[1,...,N_{POD}]}$  of U is introduced, where the brackets contain the indices of all employed modes. Hereafter, we consider that the ensemble used to determine the POD modes consists of  $N_s$  flow realizations (called time snapshots)  $U(\mathbf{x}, t_i), \mathbf{x} \in \Omega$ , taken at  $t_i \in [0, T], i = 1, ..., N_t$ .

The energetic optimality of the POD basis functions suggests that only a very small number of POD modes may be necessary to describe efficiently any flow realizations of the input data, *i.e.*  $N_r \ll N_s$ . In practice,  $N_r$  is usually determined as the smallest integer M such that the relative information content,  $RIC(M) = \sum_{i=1}^{M} \lambda_i / \sum_{i=1}^{N_s} \lambda_i$ , is greater than a predefined percentage of energy,  $\delta$ . So that  $N_{POD} = N_r$ , and the approximation (6) becomes

$$\boldsymbol{U}(\boldsymbol{x},t) \simeq \widehat{\boldsymbol{U}}^{[1,\dots,N_r]}(\boldsymbol{x},t) = \sum_{n=1}^{N_r} a_n(t) \boldsymbol{\Phi}_n(\boldsymbol{x}).$$
(7)

#### 2.2. Classical reduced order model and drawbacks

To derive a classical reduced order model only the velocity fields are used, so that  $\boldsymbol{U}(\boldsymbol{x},t) \equiv \boldsymbol{u}(\boldsymbol{x},t)$ . Thus, decomposition (7) becomes

$$\boldsymbol{u}(\boldsymbol{x},t) \simeq \sum_{n=1}^{N_r} a_n(t) \boldsymbol{\phi}_n(\boldsymbol{x}), \tag{8}$$

where  $\phi_n$  denote the velocity POD basis functions. A low dimensional dynamical system is obtained via a Galerkin projection of the Navier–Stokes equations (1). The Galerkin projection formally is written as

$$\left(\boldsymbol{\phi}_{i},\frac{\partial\boldsymbol{\boldsymbol{u}}}{\partial\boldsymbol{t}}+(\boldsymbol{\boldsymbol{u}}\cdot\boldsymbol{\nabla})\boldsymbol{\boldsymbol{u}}\right)_{\Omega}=-(\boldsymbol{\phi}_{i},\boldsymbol{\nabla}\boldsymbol{p})_{\Omega}+\left(\boldsymbol{\phi}_{i},\frac{1}{Re}\Delta\boldsymbol{\boldsymbol{u}}\right)_{\Omega}.$$
(9)

Note that since the pressure term  $(\phi_i, \nabla p)_{\Omega}$  cannot be evaluated using the standard velocity POD formulation, it is usually neglected (see discussion below). After some algebraic manipulations using decomposition (8), the reduced order model can be written as (see [19] for more details)

$$\frac{\mathrm{d}a_i(t)}{\mathrm{d}t} = \mathcal{A}_i + \sum_{j=1}^{N_r} \mathcal{B}_{ij} a_j(t) + \sum_{j=1}^{N_r} \sum_{k=1}^{N_r} \mathcal{C}_{ijk} a_j(t) a_k(t), \quad i = 1, \dots, N_r.$$
(10a)

with initial conditions

$$a_i(0) = (\boldsymbol{u}(\boldsymbol{x}, 0), \boldsymbol{\phi}_i(\boldsymbol{x}))_{\Omega}, \quad i = 1, \dots, N_r.$$
(10b)

It is well known that when Eqs. (10) are integrated in time a gradual drifting from the full-state solution to another erroneous state may arise after several vortex shedding periods, precluding a correct description of the long-term dynamics [20]. Even worse, in some cases, is that the short-term dynamics of the POD ROM may not be sufficiently accurate to be used as a surrogate model of the original high-fidelity model. Essentially, three sources of numerical errors can be identified. As it was already mentioned, the POD/Galerkin method can first present a lack of inherent numerical stability even for very simple problems [10]. Secondly, the pressure term is often neglected in the POD ROM. It is possible to model this term, but to avoid this modelisation, a pressure extended reduced order model is introduced in Section 3. The third source of instability is the truncation involved in the POD–Galerkin approach. Indeed, since only the most energetic POD modes are kept, the POD ROM is not sufficiently dissipative to prevent erroneous time amplifications of its solution. This problem is similar to that of large eddy simulation, where the energy transfers between the resolved scales and the subgrid scales have to be modelled [13]. For instance, 4 modes are sufficient to restore more than 99% of the kinetic energy of the circular cylinder wake flow (2D, laminar regime), but the solution of the such reduced order model does not converge towards the numerical solution of the Navier– Stokes equations [21]. It is thus necessary to stabilize the POD ROM. In this study, thanks to the pressure extended reduced order model, the POD ROM can be stabilized using the Navier–Stokes operator residuals evaluated with the POD flow fields reconstructions (Section 4).

#### 3. A pressure extended reduced order model

It is demonstrated that the contribution of the pressure term vanishes in many closed flows. However, Noack [11] proved that neglecting the pressure term for convectively unstable shear layers (as the mixing layer or the wake flow) can lead to large amplitude errors in the Galerkin model. A solution is to model this pressure term [11,12]. One aim of this study is to invoke the least modelisation as possible. The purpose of this section is thus to derive a pressure extended reduced order model, *i.e.* a ROM that allows to build both the velocity and the pressure fields. The pressure term can thus be easily calculated using  $p = \tilde{p}$  (see decomposition (11b)). Another key issue is that knowing the pressure field, it is possible to evaluate the Navier–Stokes residuals.<sup>3</sup> Indeed, the Navier–Stokes residuals will be used to both stabilize (Section 4) the ROM and improve (Section 5) the POD subspace.

#### 3.1. Construction of the pressure extended POD ROM

N.I

As it was mentioned in Section 2.2, reduced order modeling is based on the restriction of the weak form of the Navier– Stokes equations to the subspace  $S_{N_r}^{POD}$  spanned by the first  $N_r$  spatial eigenfunctions  $\Phi_i$ . Here, we develop a global basis for both the velocity and pressure fields (see [22] for justification and numerical demonstration). The exact flow fields  $\boldsymbol{u}$  and pare then approximated by

$$\tilde{\boldsymbol{u}}(\boldsymbol{x},t) = \sum_{i=1}^{N_r} a_i(t)\boldsymbol{\phi}_i(\boldsymbol{x}),$$

$$\tilde{p}(\boldsymbol{x},t) = \sum_{i=1}^{N_r} a_i(t)\psi_i(\boldsymbol{x}).$$
(11a)
(11b)

<sup>&</sup>lt;sup>3</sup> Only the velocity field is necessary to evaluate the residuals of the Navier–Stokes operator written in its vorticity formulation.

The velocity and the pressure basis functions,  $\phi_i$  and  $\psi_i$ , respectively, are determined using  $\boldsymbol{U}(\boldsymbol{x},t) = (\tilde{\boldsymbol{u}}(\boldsymbol{x},t), \tilde{p}(\boldsymbol{x},t))^T$  to calculate the temporal correlation tensor (3). The basis functions  $\phi_i$  and  $\psi_i$  are determined as  $\boldsymbol{\Phi}(\boldsymbol{x},t) = (\phi(\boldsymbol{x},t), \psi(\boldsymbol{x},t))^T$ ,  $\boldsymbol{\Phi}(\boldsymbol{x},t)$  being obtained from (5).

The substitution of Eqs. (11) in the Navier-Stokes momentum equations (1a) leads to

$$\sum_{j=1}^{N_r} \frac{\mathrm{d}a_j}{\mathrm{d}t} \phi_j + \left(\sum_{j=1}^{N_r} a_j \phi_j \cdot \nabla\right) \sum_{k=1}^{N_r} a_k \phi_k = -\sum_{j=1}^{N_r} a_j \nabla \psi_j + \frac{1}{Re} \sum_{j=1}^{N_r} a_j \Delta \phi_j, \tag{12}$$

that is

$$\sum_{j=1}^{N_r} \phi_j \frac{\mathrm{d}a_j}{\mathrm{d}t} + \sum_{j=1}^{N_r} \sum_{k=1}^{N_r} (\phi_j \cdot \nabla) \phi_k a_j a_k = -\sum_{j=1}^{N_r} \nabla \psi_j a_j + \frac{1}{Re} \sum_{j=1}^{N_r} \Delta \phi_j a_j.$$
(13)

A Galerkin projection of the momentum equations (13) yields

$$\left(\boldsymbol{\phi}_{i},\sum_{j=1}^{N_{r}}\boldsymbol{\phi}_{j}\frac{\mathrm{d}a_{j}}{\mathrm{d}t}+\sum_{j=1}^{N_{r}}\sum_{k=1}^{N_{r}}\left(\boldsymbol{\phi}_{j}\cdot\boldsymbol{\nabla}\right)\boldsymbol{\phi}_{k}a_{j}a_{k}+\sum_{j=1}^{N_{r}}\boldsymbol{\nabla}\psi_{j}a_{j}-\frac{1}{Re}\sum_{j=1}^{N_{r}}\Delta\boldsymbol{\phi}_{j}a_{j}\right)_{\Omega}=0.$$
(14)

The reduced order model is then

$$\sum_{j=1}^{N_r} L_{ij}^{(m)} \frac{\mathrm{d}a_j}{\mathrm{d}t} = \sum_{j=1}^{N_r} B_{ij}^{(m)} a_j + \sum_{j=1}^{N_r} \sum_{k=1}^{N_r} C_{ijk}^{(m)} a_j a_k,$$
(15)

where the coefficients<sup>4</sup>  $L_{ij}^m$ ,  $B_{ij}^m$  and  $C_{ijk}^m$  are given by

$$L_{ij}^{(m)} = + \left(\phi_i, \phi_j\right)_{\Omega},\tag{16a}$$

$$B_{ij}^{(m)} = -\left(\phi_i, \frac{1}{Re}\Delta\phi_j - \nabla\psi_j\right)_{\Omega},\tag{16b}$$

$$C_{ijk}^{(m)} = -(\phi_i, (\phi_j \cdot \nabla)\phi_k)_{\Omega}.$$
(16c)

Here, the superscript m stands for momentum equations.

In this reduced order model we used global basis functions built by POD, but this methodology could be transposed to other modal decompositions such as decomposition onto stability modes [8]. Moreover, it could be interesting to use nondivergence free modes, as Navier–Stokes residuals modes. Such modes can be used to stabilize (Section 4) and improve (Section 5) the POD ROM. Hence, if model (15) is built using nondivergence free modes, it does not satisfy the continuity equation (mass conservation). It is thus necessary to add a constraint in the reduced order model.

A modified ROM that satisfies both momentum and continuity equations can be obtained starting from the weak form of the Navier–Stokes equations

$$\left(\boldsymbol{w}_{i},\frac{\partial\boldsymbol{u}}{\partial t}+\boldsymbol{\nabla}\cdot(\boldsymbol{u}\otimes\boldsymbol{u})+\boldsymbol{\nabla}p-\frac{1}{Re}\Delta\boldsymbol{u}\right)_{\Omega}+(\boldsymbol{q}_{i},\boldsymbol{\nabla}\cdot\boldsymbol{u})_{\Omega}=\boldsymbol{0},$$
(17)

where  $\mathbf{w}_i$  and  $q_i$  belong to appropriate functional spaces. The velocity and pressure fields are expanded onto the POD basis functions  $\{\phi_i\}_{i=1}^{N_r}$  and  $\{\psi_i\}_{i=1}^{N_r}$  using Eqs. (11a) and (11b), respectively.

One approach is then to use a Galerkin projection where  $\mathbf{w}_i = \phi_i$  and  $q_i = \psi_i$  as done before (Eq. (14)). Another approach is to use  $\mathbf{w}_i = \phi_i$  and  $q_i = \alpha (\nabla \cdot \phi_i)^T$ . This choice would correspond to the minimization of the continuity residuals,  $\sum_{j=1}^{N_r} a_j \nabla \cdot \phi_j$ , in a least squares sense, so that in limit of large  $\alpha$  we have

$$\sum_{j=1}^{N_r} B^c_{ij} a_j = \mathbf{0},$$

where  $B_{ij}^{(c)} = (\nabla \cdot \phi_i)^T \nabla \cdot \phi_j$  and the superscript *c* stands for continuity equation. Numerically, this second approach gives better results, and the modified ROM that we use is thus

$$\sum_{j=1}^{N_r} L_{ij}^{(m)} \frac{\mathrm{d}a_j}{\mathrm{d}t} = \sum_{j=1}^{N_r} \left( B_{ij}^{(m)} + \alpha B_{ij}^{(c)} \right) a_j + \sum_{j=1}^{N_r} \sum_{k=1}^{N_r} C_{ijk}^{(m)} a_j a_k, \tag{18}$$

where the weight  $\alpha$  has to be fixed. In this study, we chose  $\alpha = 10^{-2}$ .

Since we use the flow-field decompositions (11), the mean flow is solved by the reduced order model. The mean flow is then  $\overline{U}(\mathbf{x}, t) = a_1(t)\Phi_1(\mathbf{x})$ . It is well known that a small drift of the first temporal coefficient  $a_1$  can occur. The flow rate is thus

<sup>&</sup>lt;sup>4</sup> In a general way, we have  $(\Phi_i, \Phi_j)_{\Omega} = \delta_{ij}$ , but not  $(\phi_i, \phi_j)_{\Omega} = \delta_{ij}$ . So,  $L_{ij}^{(m)} \neq \delta_{ij}$ .

modified. In order to keep the flow rate as constant, another constraint must be enforced in the reduced order model (18). For the 2D confined flow, the conservation of flow rate is written as

$$\int_{\mathcal{S}} u \mathrm{d}s = c, \tag{19}$$

where S is a cross-section of the channel and c is a constant. For instance, S could be the inflow or outflow height H of the channel, or, at the abscissa of the cylinder with height L, S = H - L (see Fig. 1). Numerically, the flow rate has to be constant over each slice  $S_l \equiv S(x_l), 1 \leq l \leq N_X$ , where  $N_X$  is the number of discretisation points in the *x*-direction (a cartesian mesh is used).

Denoting  $\boldsymbol{\phi} = (\phi^{u}, \phi^{v})^{T}$  and using (11a), condition (19) is approximated by<sup>5</sup>

$$\sum_{i=1}^{N_r} a_j(t) \int_{\mathcal{S}_l} \phi_j^u \mathrm{d}s = c.$$
<sup>(20)</sup>

The constant is initially evaluated by projection of a given snapshot onto the basis functions  $\phi$ . Numerically, in this study we have  $c \equiv 1$ . The flow rate conservation is written as

$$\sum_{j=1}^{N_r} \frac{\mathrm{d}a_j}{\mathrm{d}t} \int_{\mathcal{S}_l} \phi_j^u \mathrm{d}s = 0$$

Denoting by  $f_j$  the vector with components  $f_j^l = \int_{S_l} \phi_j^u ds$ , the flow rate conservation over the whole domain  $\Omega$  is written as

$$\sum_{j=1}^{N_r} \frac{\mathrm{d}a_j}{\mathrm{d}t} \boldsymbol{f}_j = \boldsymbol{0}$$

These additional constraints are now taken into account by enlarging the projection space with  $\beta f_i$ . In the limit of large  $\beta$  we have in a least square sense

$$\sum_{j=1}^{N_r} L_{ij}^r \frac{\mathrm{d}a_j}{\mathrm{d}t} = \mathbf{0},$$

where  $L_{ij}^r = f_i^T f_j$ . The superscript *r* stands for flow rate conservation. Then, the reduced order model writes

$$\sum_{j=1}^{N_r} \left( L_{ij}^{(m)} + \beta L_{ij}^{(r)} \right) \frac{\mathrm{d}a_j}{\mathrm{d}t} = \sum_{j=1}^{N_r} \left( B_{ij}^{(m)} + \alpha B_{ij}^{(c)} \right) a_j + \sum_{j=1}^{N_r} \sum_{k=1}^{N_r} C_{ijk}^{(m)} a_j a_k \tag{21a}$$

with initial conditions

$$a_i(0) = (\boldsymbol{U}(\boldsymbol{x}, 0), \boldsymbol{\Phi}_i(\boldsymbol{x}))_{\Omega}, \quad i = 1, \dots, N_r,$$
(21b)

where the weight  $\beta$  has to be fixed. In this study, we chose  $\beta = 10^2$ . This reduced order model satisfies the momentum equations, the continuity equation as well as the conservation of the flow rate, even for nondivergence free modes.

#### 3.2. Numerical results of the pressure extended POD ROM

The reduced order model (21) is tested on a 2D confined square cylinder wake flow in laminar regime (Re = 200). In this section, the POD basis  $\Phi$  is built following the POD snapshot method introduced by Sirovich [3]. Here, 80 snapshots uniformly distributed over one vortex shedding period are used to compute the discrete form of the temporal tensor (3). The corresponding eigenvalues spectrum is presented in Fig. 2. This spectrum is degenerate presenting pairs of identical eigenvalues for the fluctuating modes (the mean flow is indexed by 1). The POD basis functions are obtained via a projection of the temporal tensor eigenvectors on the whole set of snapshots. Some of them are presented in Fig. 3 in terms of iso-vorticity (noted  $\nabla_A \phi_i$ , for velocity modes  $\phi_i$ ) and isobars (for pressure modes  $\psi_i$ ). The evolution of the *RIC* introduced in Section 2.1 is presented in Fig. 4. Only the first 5 modes are sufficient to represent more than 98% of the total kinetic energy. However, another 5-modes reduced order basis containing approximatively the same percentage of energy could be derived using modes 6 and 7 instead of 4 and 5. Indeed, even if these two pairs of modes are very different (see, for instance, the topological differences between  $\phi_5$  and  $\phi_7$  in Fig. 3), they have approximatively the same energetic contribution as one can see in Fig. 5 where the individual energetic contribution (IEC) is presented. Thus, a judicious choice of the POD modes is not so evident in this case. Instead of using the RIC criterium, one can decide to keep all the fluctuating modes presenting an energy contribution greater than a given threshold (see Fig. 5). Here, all the modes with an energy contribution greater than  $10^{-2}$  are kept. This corresponds to 10 fluctuating modes plus the mean flow mode, *i.e.*  $N_r = 11$  modes.

<sup>&</sup>lt;sup>5</sup> If one uses only POD modes, we can take  $N_r = 1$  since the flow rate is only given by the mean flow. However, using other modes that do not respect *a priori* the flow rate conservation (as the residual modes),  $N_r \neq 1$ .



Fig. 2. Eigenvalues spectrum.

After having computed once the operators of the reduced order model (21) using these  $N_r = 11$  modes, a long-time flow prediction over more than 1000 vortex shedding periods is performed. Fig. 6 presents the temporal evolution of the set of coefficients  $\{a_i\}_{i=1}^{N_r}$ , solution of the system (21), over the 40 vortex shedding periods. Since the temporal tensor eigenvalues spectrum is degenerate (it presents pairs of eigenvalues), only the odd-indexed coefficients are presented (the even ones have the same behavior). It is noticeable that no divergence occurs for the long-time prediction. A comparison of the projected (projection of the Navier–Stokes solution onto the POD basis) and the predicted (solutions of the ROM (21)) limit cycles over 1000 vortex shedding periods is presented in Fig. 7. The predicted limit cycles perfectly match the projected ones even for small scales (high order modes), where the high-frequency dynamics is more complex. The good accuracy for all limit cycles also indicates that no spurious dephasing occurs between modes. The system (21) with 11 modes is numerically



Fig. 3. Representation of some POD modes. Iso-vorticity (left) and isobars (right). Dashed lines represent negative values (the pressure reference is arbitrarily chosen to be zero).



stable for short- and long-time predictions so that no calibration procedures are needed. We want to highlight that a small gradual drifting could be observed using classical POD reduced order model where the pressure term remains unmodeled. It is thus important to calculate, or at least to model, the pressure term.

As it was already mentioned (Section 2.2), when system (21) is integrated in time a gradual drifting from the full-state solution to another erroneous state may arise after several vortex shedding periods if only a very small number of modes are kept. Indeed as it was shown in Figs. 8 and 9, the solution of model (21) built with 5 modes reaches erroneous limit cycles, and can even diverge with 3 modes (see Figs. 10 and 11). In this simple test case, only  $N_r = 11$  modes are sufficient to build a stable ROM. However, in many practical applications (three-dimensional flows, turbulent regimes, complex geome-



Fig. 6. Temporal evolutions of the predicted POD coefficients over 40 vortex shedding periods. Eleven modes model.



**g. 7.** Comparison of the projected (NS:  $\diamond$ ) and the predicted (ROM: -) limit cycles over 1000 vortex shedding periods. Eleven modes model.



number of POD modes that represent 99% of the total kinetic energy is large. Usually, approximatively 60–80% energy can be retained, so that reduced order models are unstable. The following section presents methods vavier–Stokes residuals to stabilize reduced order models built with a very low number of modes (namely n our case).

#### of reduced order models

errors due to the truncation involved in the POD–Galerkin approach, different kinds of POD ROM/Galerin hods are commonly used.

of stabilization methods uses eddy viscosity. Since the early works on POD ROM, it was shown that artificial stabilization [23]. A natural way is to add a constant viscosity acting the same way on all POD modes: this erg model [24,25]. The global dimensionless viscosity 1/Re is thus replaced by another one defined as oblem is then to determine or to adjust the constant c > 0 in order to obtain a better accuracy for the r and Fasel [26] and Rempfer [27] have improved this idea by supposing that the dissipation is not iden-POD modes. Thus, the global viscosity could be replaced by modal viscosities  $1/Re_i = (1 + c_i)/Re$  on each then necessary to determine a set of correction coefficients spanned by  $c_i$  for  $i = 1, ..., N_r$ . In [27], it is ddy viscosities are a function of the coupled modes index  $j = 1, ..., N_r/2$ . The coefficients  $c_j$  are such that s the unique constant to determine or to adjust. More recently, Karniadakis employed a dissipative model hing viscosity model (SVVM) to formulate alternative stabilization approaches [28] and to improve the r models [20]. In this spirit, an optimal spectral viscosity model based on parameters identification techosed [29].

f stabilization methods consists in calibrating the polynomial coefficients of the POD model [12,30–32]. tensor *B* are determined using a least square or an adjoint method so that the predicted coefficients ossible to the eigenvectors of the temporal correlation tensor (see Eq. (2)). These calibration methods, tification, are very similar to spectral viscosity closures. However, calibration methods allow such a inter-modal transfers.

abilization methods uses a penalty term. This consists in introducing a new term in the reduced order and Cazemier et al. [34] used modal kinetic equations to determine viscosities to be added on each 3] supposes that the lack of interaction between the calculated and the nonresolved modes is responence of the temporal POD coefficients. To solve this problem, another artificial linear coefficient is lynamical system. The POD ROM is then written as

 $i + \sum_{i=1}^{N_r} \sum_{k=1}^{N_r} \mathcal{C}_{ijk} a_j a_k + \mathcal{H}_i a_i,$ 

PO tica POI argi  $C_i =$ calle accu nique Th All the  $a_i(t)$  ar based of represe The t model. C POD mod sible of a introduced

vi

is

(1)

dai dt

We showed that the POD ROM is stable if a sufficient number of modes is taken into account in the model. For instance, model  $A^{(11)}$  is stable. Thus, the POD ROM is stable if the POD flow fields get close enough to the exact flow field. The exact flow field is written as

$$u(x,t) = \tilde{u}(x,t) + u'(x,t),$$

$$p(x,t) = \tilde{p}(x,t) + p'(x,t),$$
(23a)
(23b)

where u' and p' denote the fine scales that are not resolved by the POD ROM. Unfortunately, the exact resolution of the fine scales equations (with solutions  $\mathbf{u}'$  and  $\mathbf{p}'$ ) requires computational costs similar to those required for solving the complete Navier-Stokes equations. The objective of the following section is thus to derive stabilization methods that make use of approximations of these fine scales.

#### 4.1. Residuals-based stabilization method: model $B^{[N_r;K]}$

The goal of this method is to approximate the fine scales u' and p' onto some adapted basis functions. The exact flow field is hence approximated as

$$\boldsymbol{u}(\boldsymbol{x},t) = \underbrace{\sum_{i=1}^{N_r} a_i(t)\boldsymbol{\phi}_i(\boldsymbol{x})}_{\tilde{\boldsymbol{u}}(\boldsymbol{x},t)} + \underbrace{\sum_{i=N_r+1}^{N_r+K} a_i(t)\boldsymbol{\phi}_i'(\boldsymbol{x})}_{\boldsymbol{u}'(\boldsymbol{x},t)},$$
(24a)

$$p(\mathbf{x},t) = \underbrace{\sum_{i=1}^{N_r} a_i(t)\psi_i(\mathbf{x})}_{p(\mathbf{x},t)} + \underbrace{\sum_{i=N_r+1}^{N_r+K} a_i(t)\psi_i'(\mathbf{x})}_{p'(\mathbf{x},t)}.$$
(24b)

If the basis functions  $\phi'_i$  and  $\psi'_i$  are, respectively, equal to  $\phi_i$  and  $\psi_i$ , the energetic representation is improved. Indeed, the more POD modes are retained, the better is the energetic representation. This is a classical remark of POD. However, it is not granted that POD modes are optimal to stabilize the ROM. If we suppose that a sufficient amount of energy is captured with  $N_r$  modes, it is not necessary to add any more energy in the POD ROM, but rather some viscous dissipation. For instance, model  $A^{[5]}$ , *i.e.* model  $B^{[3;2]}$  with  $\Phi' \equiv \Phi$  in (24), is not stable. However, it will be demonstrated that another 5-modes model with  $N_r = 3$  and K = 2 are stable using different modes  $\Phi'$ . It is well known that the residuals of the governing equations play a major role to stabilize dynamical systems [14]. The leading idea of this section is thus to take  $\phi'_i$  and  $\psi'_i$  as being the POD basis functions of the POD-NS residuals defined by (22). The method is the following.

#### Algorithm 1 (Residuals-based stabilization).

- (1) Integrate the ROM  $A^{[N_r]}$  to obtain  $a_i(t)$  and compute  $N_s$  coefficients  $a_i(t_k)$ ,  $k = 1, ..., N_s$ . (2) Compute the POD flow fields  $\tilde{\boldsymbol{u}}(\boldsymbol{x}, t_k) = \sum_{i=1}^{N_r} a_i(t_k) \phi_i(\boldsymbol{x})$ ,  $\tilde{p}(\boldsymbol{x}, t_k) = \sum_{i=1}^{N_r} a_i(t_k) \psi_i(\boldsymbol{x})$ , and then the POD-NS residuals  $\boldsymbol{R}_{M}(\boldsymbol{x},t_{k})$  and  $R_{C}(\boldsymbol{x},t_{k})$ .
- (2) Compute the POD modes  $\phi'_i(\mathbf{x})$  and  $\psi'_i(\mathbf{x})$  of the POD-NS residuals  $\mathbf{R}_M(\mathbf{x}, t_k)$  and  $R_C(\mathbf{x}, t_k)$ .
- (3) Add the K first residual modes  $\phi'_i$  and  $\psi'_i$  to the existing POD basis  $\phi_i$  and  $\psi_i$  (using Gram–Schmidt process) and built a new ROM (here the mass and flow rate constraints are important).

The reduced order model obtained with this algorithm is noted as  $B^{[N_r;K]}$ , where  $N_r$  is the number of initial POD basis functions and K is the number of POD residuals modes. The results of this model are presented in Section 4.3.

### 4.2. SUPG and VMS methods: models $C^{[N_r]}$ and $D^{[N_r]}$

The streamline upwind Petrov–Galerkin (SUPG) and the variational multiscale (VMS) methods are devised to provide appropriate modeling and stabilizations for the numerical solution of the Navier-Stokes equations. The SUPG method is a simplified version of the complete VMS method, and the main steps leading to these models are described in [14].

The main idea of both SUPG and VMS methods is to approximate the fine scales by

$$\begin{aligned} \mathbf{u}' \simeq -\tau_M \mathbf{R}_M \\ p' \simeq -\tau_C \mathbf{R}_C, \end{aligned} \tag{25a}$$

where  $\tau_M$  and  $\tau_c$  denote some constants to be fixed.

The SUPG and VMS reduced order models can be formally written as

$$\sum_{j=1}^{N_r} \left( L_{ij}^{(m)} + \beta L_{ij}^{(r)} \right) \frac{\mathrm{d}a_j}{\mathrm{d}t} = \sum_{j=1}^{N_r} \left( B_{ij}^{(m)} + \alpha B_{ij}^{(c)} \right) a_j + \sum_{j=1}^{N_r} \sum_{k=1}^{N_r} C_{ijk}^{(m)} a_j a_k + F_i(t),$$
(26)

where the "penalization" term  $F_i(t)$  is defined as follows:

• For the SUPG reduced order model, noted as  $C^{[N_r]}$ , we have

$$F_{i}^{SUPG}(t) = (\tilde{\boldsymbol{u}} \cdot \nabla \boldsymbol{\phi}_{i} + \nabla \psi_{i}, \tau_{M} \boldsymbol{R}_{M}(\boldsymbol{x}, t))_{\Omega} + (\nabla \cdot \boldsymbol{\phi}_{i}, \tau_{C} \boldsymbol{R}_{C}(\boldsymbol{x}, t))_{\Omega}.$$
(27)

#### Table 1

Brief description of the different ROMs.  $N_r$  denotes the number of the retained POD modes.

ROM	Method
$\begin{array}{l} A^{[N_r]} \\ B^{[N_r,K]} \\ C^{[N_r]} \\ D^{[N_r]} \end{array}$	No stabilization Stabilization with K residuals POD modes SUPG stabilization VMS stabilization

#### • For the VMS reduced order model, noted as $D^{[N_r]}$ , we have

$$F_{i}^{\text{VMS}}(t) = F_{i}^{\text{SUPG}}(t) + (\tilde{\boldsymbol{u}} \cdot (\boldsymbol{\nabla} \boldsymbol{\phi}_{i})^{T}, \tau_{M} \boldsymbol{R}_{M}(\boldsymbol{x}, t))_{\Omega} - (\boldsymbol{\nabla} \boldsymbol{\phi}_{i}, \tau_{M} \boldsymbol{R}_{M}(\boldsymbol{x}, t) \otimes \tau_{M} \boldsymbol{R}_{M}(\boldsymbol{x}, t))_{\Omega}.$$

$$(28)$$

The parameters  $\tau_M$  and  $\tau_c$  can be found using some scaling arguments (see [39] for more details), so that no modelisation is required. This can lead to an universal POD ROM closure model. However, in what follows, parameters  $\tau_M$  and  $\tau_c$  are determined using an optimal formulation, so that the temporal coefficients  $a_i(t)$  fit as best as possible the eigenvectors of the correlations tensor (3).

For clarity reasons, Table 1 summarizes the different ROMs introduced above, where  $N_r$  denotes the number of POD modes used to compute the ROMs.

#### 4.3. Results of stabilization methods

In this section, we take exactly the same configuration and parameters than those used in Section 3.2. The confined square cylinder wake flow for Re = 200 and the ROMs are built using  $N_s = 80$  snapshots uniformly distributed over one vortex shedding period. The model based on residual modes,  $B^{[N_r;K]}$ , is integrated following Section 4.1. The SUPG model,  $C^{[N_r]}$ , and the VMS model,  $D^{[N_r]}$ , are integrated using (27) and (28), respectively. The unstable configurations observed in Section 3.2 are studied, *i.e.*  $N_r = 5$  and  $N_r = 3$ . Moreover, for model  $B^{[N_r;K]}$ , only K = 2 residual modes are used. These two additional modes present a different behavior than POD modes 4 and 5 used to build model  $A^{[5]}$  (see Fig. 12).

The temporal evolutions of the predicted POD coefficients  $a_i(t)$  obtained with models *B*, *C* and *D*, over 40 vortex shedding periods, are presented in Figs. 13 and 15 for  $N_r = 5$  and  $N_r = 3$ , respectively. The three reduced order models provide an accurate description of the asymptotic attractor (compared with the results in Fig. 6), and results are indistinguishable between models *B*, *C* and *D*. The limit cycles obtained with models *B*, *C* and *D* are represented in Figs. 14 and 16 for  $N_r = 5$  and  $N_r = 3$ , respectively. These limit cycles represent 1000 vortex shedding periods. There are no differences between the results obtained from the three models. These limit cycles are compared to the exact ones obtained by DNS (projection of the snapshots onto the POD basis) in the figures. Excellent agreements are observed between all these limit cycles, thus validating all the stabilization methods described in Sections 4.1 and 4.2.

In order to highlight the differences between the stabilization methods presented above, we study the  $L^2$  norm of the POD-NS residuals introduced in (22). Figs. 17(a) and 18(a) show the temporal evolutions of the  $L_2$  norms of the POD-NS residuals obtained initial model *A* (obtained from (21) or (26) with  $F_i = 0$ ) with  $N_r = 5$  and  $N_r = 3$ , respectively. For the sake of clarity, only 20 vortex shedding periods are represented. An initial growth and then an asymptotic limit is reached with  $N_r = 5$ . This can be explained by the fact that the dynamic converges towards another attractor (see Fig. 7). On the contrary, an exponential divergence occurs with  $N_r = 3$  (see also the divergence in Fig. 11). We compare the effectiveness of models *B*, *C* and *D* in Figs. 17(b) and 18(b). All stabilized reduced order models are accurate (low values of the POD-NS residuals norm). For  $N_r = 5$  or  $N_r = 3$ , models *C* and *D* are more accurate than model *B* (lower residuals). Moreover, the VMS model,



Fig. 12. Comparison between original POD modes (left, model A<sup>[5]</sup>) and residuals modes (right, model B<sup>[3:2]</sup>). Dashed lines represent negative values.





**Fig. 16.** Comparison of the projected (NS:  $\Diamond$ ) and the predicted (ROM: –) limit cycles over 1000 vortex shedding periods. Three modes model with stabilization. (Indistinguishable difference between models  $B^{[3,2]}, C^{[3]}$  and  $D^{[3]}$ .)







**Fig. 19.** Comparison of the iso-vorticity representation of some POD modes. The initial POD basis at  $Re_1 = 100$  (left) and the target POD basis at  $Re_2 = 200$  (right). Dashed lines represent negative values.

The aim of this section is thus to present efficient methods to improve – or actualize – the functional subspace. As it was mentioned, the underlying idea is to adapt the basis when input system parameters change (Reynolds number, control parameters, *etc.*). For simplicity reasons, we will only focus on Reynolds number modifications, but the forthcoming process is easily transposable to other parameters modifications. Here, for instance, our goal is to obtain the target basis built at  $Re_2 = 200$  starting from the initial basis built at  $Re_1 = 100$ . These two basis are quite different, especially for the mean flow mode  $\phi_1$  (see Fig. 19).

In what follows two methods are considered.

- The objective of the first method is to improve the basis using residuals-based approximations of the missing fields u' and p'. This is exactly the same idea that was used in (Section 4.1), applied iteratively (Section 5.1).
- The aim of the second method is to modify the POD database by means of DNS simulations (Section 5.2).

These two methods are based on an iterative process. The current POD basis obtained during the actualization process is simply denoted by  $\mathbf{\Phi} \equiv \mathbf{\Phi}^{(n)}$  where *n* is the number of iterations considered (number of POD basis actualization cycles). Obviously,  $\mathbf{\Phi}^{(0)} = \mathbf{\Phi}^{Re_1}$ , and we expect at least that  $\mathbf{\Phi}^{(+\infty)} = \mathbf{\Phi}^{Re_2}$ . The convergence criterion tested in this study is the best projection of  $\mathbf{\Phi}^{(n)}$  onto  $\mathbf{\Phi}^{Re_2}$ , noted  $\mathbf{\Phi} \cdot \mathbf{\Phi}^{Re_2}$ . For our Navier–Stokes test case with  $Re_1 = 100$  and  $Re_2 = 200$ , the initial value of the convergence criterion is  $\mathbf{\Phi}^{Re_1} \cdot \mathbf{\Phi}^{Re_2} \approx 0.5$ . The expected final value is naturally  $\mathbf{\Phi}^{(n)} \cdot \mathbf{\Phi}^{Re_2} \approx 1$ , with the smallest possible number of iterations *n*.

#### 5.1. A Krylov-like method

The first method used to improve the functional subspace uses successive application of the Navier–Stokes operator on the residuals. This method is nothing but an iterative version of the stabilization process introduced in Section 4.1. We have seen that this method does a good job to stabilize POD ROM, so it is reasonable to investigate its performance to improve the functional subspace. This subspace adaptation is described by the following algorithm, and is schematically represented in Fig. 20. The numerical integration of the ROM (21) is always performed at the target Reynolds number  $Re \equiv Re_2$  (the Reynolds number Re is required to evaluate  $B^{(m)}$ , see equation (16b)). Otherwise, all the POD coefficients, *i.e.*  $L^{(m)}$ ,  $L^{(r)}$ ,  $B^{(m)}$ ,  $B^{(c)}$ ,  $C^{(m)}$ , are built using the current updated POD basis,  $\Phi = \Phi^{(n)}$ , following (16c).



**Algorithm 2** (*Krylov like adaptation method*). Start with the POD basis to be improved,  $\Phi_i$  with  $i = 1, ..., N_r$  (in our case built for  $Re = Re_1$ ). Let  $N_0 = N_r$  and  $\mathcal{T} = [0, T]$  be an observation period.

- (1) Build and solve the corresponding ROM over  $\mathcal{T}$  with  $Re = Re_2$  to obtain  $a_i(t)$  and extract  $N_s$  snapshots  $a_i(t_k)$  with  $i = 1, ..., N_r$  and  $k = 1, ..., N_s$ . Compute the flow fields  $\tilde{\boldsymbol{u}}(\boldsymbol{x}, t_k)$ ,  $\tilde{\boldsymbol{p}}(\boldsymbol{x}, t_k)$  from (11).
- (2) Compute the POD-NS residuals  $\boldsymbol{R}(\boldsymbol{x}, t_k) = (\boldsymbol{R}_M(\boldsymbol{x}, t_k), R_C(\boldsymbol{x}, t_k))^T$  from (22).
- (3) Compute the POD modes  $\tilde{\Phi}(\mathbf{x}) = (\tilde{\phi}(\mathbf{x}), \tilde{\psi}(\mathbf{x}))^T$  from the POD-NS residuals database  $\mathbf{R}(\mathbf{x}, t_k)$ ,  $k = 1, ..., N_s$ .
- (4) Add the *K* firsts residual modes  $\tilde{\Phi}(\mathbf{x})$  to the existing POD basis  $\Phi_i(\mathbf{x})$  (using Gram–Schmidt process)
  - $\Psi + \Phi \to \Phi \quad \bullet$ 
    - $N_r \leftarrow N_r + K$
  - If  $N_r$  is below than a threshold,  $N_{max}$ , return to 1. Else, go to 5.
- (5) Do step (1). From fields  $\tilde{u}$  and  $\tilde{p}$ , perform a new POD compression from with  $N_r = N_0$ .
  - If a convergence criterion is satisfied, stop. Else, return to 1.

This algorithm is a simplified version of a generalized minimal residual (GMRES) algorithm for linear system (see [48] for more details about GMRES method).

Before testing this adaptation method on the two-dimensional confined square cylinder wake flow governed by the Navier–Stokes equations (Section 5.1.2), a simple one-dimensional test case is performed on the Burgers equation (Section 5.1.1).

5.1.1. A one-dimensional test case: the Burgers equation The Burgers equation, in its dimensionless form, is written as

$$\mathcal{L}_B(u) = \frac{\partial u}{\partial t} + \frac{1}{2} \frac{\partial u^2}{\partial x} - \frac{1}{Re} \frac{\partial^2 u}{\partial x^2} = 0, \tag{29}$$

with an initial condition<sup>6</sup>

$$u(x,0) = \sin\left(\pi \frac{\tan(c_s(2x-1))}{\tan(c_s)}\right) \quad \text{and} \quad c_s = 1.3$$
(30)

and boundary conditions

$$u(0,t) = 0,$$
  
 $u(L,t) = 0.$  (31)

This equation is solved onto the domain  $\mathcal{D}$  defined by

 $\mathcal{D} = \{ (x, t) \in [0, 1] \times [0, 1] \}.$ 



**Fig. 21.** Comparison of few POD basis functions obtained for  $Re_1 = 50$  (-) and  $Re_2 = 300$  (---).

We chose the initial and target POD basis,  $\Phi^{Re_1}$  and  $\Phi^{Re_2}$ , such that  $\Phi^{Re_1} \cdot \Phi^{Re_2} \approx 0.5$ , *i.e.*  $Re_1 = 50$  and  $Re_2 = 300$  in this context. We used 40 snapshots uniformly distributed over the whole observation domain T = [0, 1]. Fig. 21, shows a comparison between few POD basis functions obtained for  $Re_1$  and  $Re_2$ . Significant differences are observable. Results of the adaptation process are presented in Fig. 22 which shows the evolution of the convergence criterion versus the number of iterations. Convergence is obtained in less than 6 iterations. In other words, the POD basis at  $Re_2$  is obtained from that at  $Re_1$  using only 6 integrations of the POD ROM, without any DNS. The POD basis functions at  $Re_2$  can even be determined starting from one mode (the normalized initial condition), or from any given basis. This represents a very efficient method to actualize a POD basis for the 1D Burgers equation. It is then of interest to see if this adaptation method can provide as good results for the Navier–Stokes equations.

#### 5.1.2. The confined square cylinder wake flow

For the confined square cylinder wake flow (2D Navier–Stokes equations), we take  $Re_1 = 100$ ,  $Re_2 = 200$  and we also use 40 snapshots distributed uniformly over one vortex shedding period *T*, so that T = [0, T], of the 2D confined square cylinder



Fig. 22. Evolution of the convergence criterion versus the number of iterations for the 1D burgers equation.



Fig. 23. Evolution of the convergence criterion versus the number of iterations for the 2D Navier-Stokes equations.

wake flow. During the adaptation process, the vortex shedding period *T* has to be actualized in consequence. Results are presented in Fig. 23. Unfortunately, no convergence is obtained. The algorithm is stopped when the computational costs were estimated to be larger than those necessary using only DNS. Same results are obtained using 400 snapshots uniformly distributed over an observation period  $\mathcal{T} = [0, 10T]$ , not presented here. The information contained into the Navier–Stokes residuals are not sufficient to improve the POD basis functions for a dynamical change, but they are sufficient to stabilize the ROM for a given dynamic (see Section 4.1). One possible explanation is that the approximation of the missing scales,  $\boldsymbol{u}'(\boldsymbol{x},t) = -\tau_M \boldsymbol{R}_M(\boldsymbol{x},t)$  and  $p'(\boldsymbol{x},t) = -\tau_C R_C(\boldsymbol{x},t)$ , is only valid for fine scales, *i.e.* the ones that are not represented due to the truncation of the POD basis. Residual modes have just dissipative behavior. We can ask if it is possible to find good values for parameters  $\tau_M$  and  $\tau_C$  for "quite large" missing scale. The answer is not so clear yet. One solution could be to look for  $\boldsymbol{U}'(\boldsymbol{x},t) = \boldsymbol{M}(t)\boldsymbol{R}(\boldsymbol{x},t)$ , where  $\boldsymbol{M} \in \mathbb{R}^{3\times3}$  for the 2D Navier–Stokes equations.

A step toward the full GMRES algorithm was also made. Few Arnoldi modes,  $\Phi''_n = A\Phi'_n, \Phi'''_n = A\Phi''_n, \dots$ , have been added to the initial basis. The operator *A* denotes the linear operator obtained after an adapted discretization of the Navier–Stokes equations, so that  $A\mathbf{U} = b(\mathbf{U})$ . The results have not been really improved. A large Arnoldi basis should be certainly necessary, but the numerical costs required to generate (and to solve) the POD ROM become prohibitive.

The following section presents another kind of algorithm that couples POD ROM with DNS. If the Navier–Stokes solution lives on the same attractor (no dynamic change), the DNS can be greatly accelerated using POD basis functions. Indeed, a Galerkin free reduced order model is recently used as DNS accelerator [49].

#### 5.2. A hybrid DNS/POD ROM method

It has been demonstrated that the percentage of the reconstruction energy decreases rapidly outside the temporal interval defined by the snapshots database for three-dimensional flows [30]. Thus, the aim of the present methodology is to update the database statistics when time evolves. By doing this, the POD basis is actualized and represents with high fidelity the current flow. The main idea is to implement a process allowing to replace older snapshots with new ones at low numerical costs. These new snapshots can be obtained using few DNS iterations. Once a new POD basis is available, a new ROM is constructed and integrated until a new snapshot is needed. Then, the process is repeated. We chose to take snapshot periodically. A schematic representation of the algorithm is presented in Fig. 24. Let us denote  $Re_1$  the Reynolds number used to build the initial POD subspace, and  $Re_2$  the Reynolds number associated to the new desired dynamic. All the simulations (ROM and DNS in Fig. 24) are performed at  $Re_2$ . After few DNS iterations, a new snapshot is available. To build a new POD subspace, we have to update the database (Section 5.2.1) to compute the correlations tensor (Section 5.2.2) and to compute the new POD basis (Section 5.2.3). All the coefficients of the ROM (21), *i.e.*  $L^{(m)}$ ,  $B^{(m)}$ ,  $B^{(c)}$ ,  $C^{(m)}$ , are built using the current POD basis (corresponding to the modified database), and the integration is performed using  $Re \equiv Re_2$ . Indeed, the Reynolds number Re is required to evaluate  $B^{(m)}$  (see(16b)). In order to obtain a "real time" POD basis actualization, efficient methods are developed for each step.



Fig. 24. Schematic representation of the hybrid method DNS-POD ROM to improve the POD subspace.

#### 5.2.1. Database modification

Following equation (7) each snapshot included in the POD database can be approximated by

$$\widehat{\boldsymbol{U}}^{[1,\dots,N_r]}(\boldsymbol{x},t_k) = \sum_{n=1}^{N_r} a_n(t_k) \boldsymbol{\Phi}_n(\boldsymbol{x}).$$
(32)

Using one new snapshot, the POD database is modified either by adding this new snapshot, or by replacing the older snapshot by this new one. The position of the new snapshot in the database is denoted by *s*. The number of snapshots included in the new database is thus  $N = \max(s, N_s)$ . The snapshot  $U_s$  can be written as the sum of its projection onto the POD basis plus its orthogonal part as

$$\widetilde{\boldsymbol{U}}(\boldsymbol{x},t_s)^{[1,\ldots,N_r]} = \widehat{\boldsymbol{U}}^{[1,\ldots,N_r]}(\boldsymbol{x},t_s) + \boldsymbol{U}_s^{\perp}(\boldsymbol{x},t_s).$$

Therefore, in general, each snapshot is

$$\widetilde{\boldsymbol{U}}(\boldsymbol{x},t_k)^{[1,\dots,N_r]} = \widehat{\boldsymbol{U}}^{[1,\dots,N_r]}(\boldsymbol{x},t_k) + \delta_{ks} \boldsymbol{U}^{\perp}(\boldsymbol{x},t_s),$$

where  $\delta$  represents the Kronecker symbol.

#### *5.2.2. Evaluation of the temporal correlation tensor*

In order to reduce the computational costs, the temporal correlation tensor *C* is calculated using the snapshots decomposition  $\widetilde{U}_{k}^{[1...,N_{r}]}$ .

$$\begin{split} \mathcal{C}(t_k, t_l) &= \left(\widetilde{\boldsymbol{U}}(\boldsymbol{x}, t_k)^{[1, \dots, N_r]}, \widetilde{\boldsymbol{U}}(\boldsymbol{x}, t_l)^{[1, \dots, N_r]}\right)_{\Omega} = \left(\sum_{i=1}^{N_r} a_i(t_k)\phi_i(\boldsymbol{x}) + \delta_{ks}\boldsymbol{U}^{\perp}(\boldsymbol{x}, t_k), \sum_{j=1}^{N_r} a_j(t_l)\phi_j(\boldsymbol{x}) + \delta_{ls}\boldsymbol{U}^{\perp}(\boldsymbol{x}, t_l)\right)_{\Omega} \\ &= \sum_{i=1}^{N_r} \sum_{j=1}^{N_r} a_i(t_k)a_j(t_l)\underbrace{\left(\phi_i(\boldsymbol{x}), \phi_j(\boldsymbol{x})\right)_{\Omega}}_{=\delta_{ij}} + \delta_{ks}\delta_{ls}\left(\boldsymbol{U}^{\perp}(\boldsymbol{x}, t_k), \boldsymbol{U}^{\perp}(\boldsymbol{x}, t_l)\right)_{\Omega} + \delta_{ls}\sum_{i=1}^{N_r} a_i(t_k)\underbrace{\left(\phi_i(\boldsymbol{x}), \boldsymbol{U}^{\perp}(\boldsymbol{x}, t_l)\right)_{\Omega}}_{=0} \\ &+ \delta_{ks}\sum_{j=1}^{N_r} a_j(t_l)\underbrace{\left(\boldsymbol{U}^{\perp^T}(\boldsymbol{x}, t_k)\phi_j(\boldsymbol{x})\right)_{\Omega}}_{=0}. \end{split}$$

Hence, the approximation of the temporal correlation tensor is simply written as

$$C(t_k, t_l) = \sum_{i=1}^{N_r} a_i(t_k) a_i(t_l) + \delta_{ks} \delta_{ls} \int_{\Omega} \sum_{i=1}^{n_c} U^{\perp i}(\mathbf{x}, t_k) U^{\perp i}(\mathbf{x}, t_l) d\mathbf{x}.$$
(33)

The evaluation of this matrix is very fast. Indeed, it is not necessary to calculate snapshots correlations on the whole mesh, but only correlations on the retained temporal coefficients. Other than that, it is only necessary to evaluate one auto-correlation with  $U_s^{\perp}$  to evaluate the component  $C_{ss}$ .

#### 5.2.3. Actualization of the POD basis

All the quantities evaluated before the calculation of the temporal correlation tensor (33) are superscripted by (n). The actualized POD basis could be evaluated from (see [3] for more details)

$$\begin{split} \boldsymbol{\phi}_{k}^{(n+1)}(\boldsymbol{x}) &= \frac{1}{\lambda_{k}^{(n+1)}} \sum_{j=1}^{N_{r}} \widetilde{\boldsymbol{U}}^{(n)}(\boldsymbol{x},t_{j}) a_{k}^{(n+1)}(t_{j}) = \frac{1}{\lambda_{k}^{(n+1)}} \sum_{j=1}^{N_{r}} \left( \sum_{i=1}^{N_{r}} a_{i}^{(n)}(t_{j}) \boldsymbol{\phi}_{i}^{(n)}(\boldsymbol{x}) + \delta_{js} \boldsymbol{U}^{\perp(n)}(\boldsymbol{x},t_{j}) \right) a_{k}^{(n+1)}(t_{j}) \\ &= \frac{1}{\lambda_{k}^{(n+1)}} \sum_{i=1}^{N_{r}} \sum_{j=1}^{N_{r}} a_{k}^{(n+1)}(t_{j}) a_{i}^{(n)}(t_{j}) \boldsymbol{\phi}_{i}^{(n)}(\boldsymbol{x}) + \frac{1}{\lambda_{k}^{(n+1)}} \boldsymbol{U}^{\perp(n)}(\boldsymbol{x},t_{s}) a_{k}^{(n+1)}(t_{s}). \end{split}$$

Introducing the modal correlation tensor  $K^{(n+1)} \in \mathbb{R}^{N_s \times N_s}$  between old and new time-dependent POD eigenfunctions,

$$K_{ki}^{(n+1)} = \frac{1}{\lambda_k^{(n+1)}} \sum_{j=1}^{N_r} a_k^{(n+1)}(t_j) a_i^{(n)}(t_j),$$

and the vector

$$\boldsymbol{S}_{k}^{(n+1)}(\boldsymbol{x}) = \frac{1}{\lambda_{k}^{(n+1)}} \boldsymbol{U}^{\perp(n)}(\boldsymbol{x}, t_{s}) \boldsymbol{a}_{k}^{(n+1)}(t_{s}),$$

we have

$$\phi_k^{(n+1)}(\mathbf{x}) = \sum_{i=1}^{N_r} K_{ki}^{(n+1)} \phi_i^{(n)}(\mathbf{x}) + \mathbf{S}_k^{(n+1)}(\mathbf{x}).$$
(34)

534





Fig. 27. Evolution of the convergence criterion versus the number of vortex shedding periods.



Fig. 28. Schematic explanation for the convergence of the hybrid DNS-POD method. ... exact "target"; --- POD ROM: - DNS.

from  $Re_1 = 100$  to  $Re_2 = 200$  is presented. As in the previous illustration, a 40 snapshots database is used to compute the initial POD basis  $\phi^{(0)}$ .

We consider the results concerning the hybrid method. The initial and target POD basis,  $\Phi^{Re_1}$  and  $\Phi^{Re_2}$ , correspond to Reynolds numbers  $Re_1 = 100$  and  $Re_2 = 200$ , respectively. Fig. 27 shows the evolution of the convergence criterion  $\Phi \cdot \Phi^{Re_2}$  versus the number of vortex shedding periods (and so versus the number of actualization iterations) for different percentages of actual the DNS. Denoting  $T_{NS}$  and  $T_{ROM}$  the time intervals where we use either DNS or POD ROM,<sup>7</sup> respectively, the percentage of DNS is defined as  $P_{NS} = T_{NS}/(T_{NS} + T_{ROM})$ .

<sup>&</sup>lt;sup>7</sup> We do not use the number of iterations because the POD ROM allows to use greater time steps than the DNS.

The simulation with  $P_{NS} = 100\%$  does not involve any ROM. It can be seen that 10 vortex shedding periods are necessary to converge towards the target POD basis using only DNS. This time corresponds to that required for perturbations, coming from the dynamics changes  $Re_1 \rightarrow Re_2$ , to cross the whole simulation domain  $\Omega$  and to reach an asymptotic behavior. The same results can approximatively be obtained with  $P_{NS} = 90\%$ ,  $P_{NS} = 80\%$  and with less accuracy 70% DNS. However, no convergence is obtained with less than  $P_{NS} = 70\%$ . Hence, a sufficient amount of DNS is necessary to converge toward the target POD basis. An explanation of this phenomenon is given in Fig. 28. Since the POD basis is not well adapted to give a good representation of the current flow (there is a delay in the adaptation process), the solution of the POD ROM moves away from the exact "desired" solution. The DNS has to be able to correct this drift. If the DNS is able to correct the error made by the POD ROM (see Fig. 28(a) and (b)), the convergence is obtained. However, if the DNS is not able to correct this error (see Fig. 28(c)), it is not possible to converge toward the target basis.

A few comments about the numerical costs can be given. The computational cost due to the ROM is negligible in comparison with the DNS ones. Also, the POD basis update described in Sections 5.2.1, 5.2.2 and 5.2.3 generates very low numerical costs. In practice, our estimation is that taking  $P_{NS} = 70\%$ , which still gives good results in terms of approximation, approximatively 20% of the total numerical costs can be saved as compared to DNS.

#### 6. Conclusions

The objective of this paper was to improve reduced order modeling based on proper orthogonal decomposition. Indeed, the proper orthogonal decomposition method is a viable technique to build low dimensional models, but it also presents several drawbacks: (i) since usually only velocity fields are used to build the reduced order model, the pressure term has to be modeled, (ii) it is necessary to model the effects of the fine scale that are not explicitly taken into account in the expansion and that are responsible of the main part of the viscous dissipation, and (iii) the POD basis functions are only able to give a good representation of the flow dynamics included in the given snapshots database.

We have shown how to build a pressure extended reduced order model at no additional cost in comparison with a standard velocity ROM. Using this pressure extended ROM, it is not necessary to model the pressure term. Since both the velocity and pressure POD fields are available, it is possible to evaluate the Navier–Stokes operator residuals using these POD fields. Although this model gives very good results in term of asymptotic solution, it is still necessary to model the effects of the unresolved fines scales.

In this respect, different stabilization methods for the reduced order model were taken under consideration. The stabilization we propose consists in modeling the effect of the missing fine scales. To this end, we made the choice of using residuals of Navier–Stokes operator evaluated from POD fields. The first method proposed consists in enlarging the POD subspace with few residuals modes. If the original number of modes is not large enough to provide appropriate dissipation, this method is able to stabilize the model. The stabilization is not due to the enlargement of the POD subspace, in fact an unstable model can be stabilized replacing a few number of original POD modes with the same number of residual modes. No empiric parameter has to be estimated in this approach. The second approach proposed relies on an approximation of the fine scale equation. Both SUPG and VMS methods give good results. In this approach, only two parameters have to be estimated. In some specific cases [39], these parameters can be approximated using some scaling arguments, leading to an universal model with no empiricism.

Finally, we have tried to improve the POD functional subspace. The goal was to derive efficient methods to adapt the POD basis when dynamics changes (with control parameters). The first method is a Krylov-like method. This method is based on iteratively including in the basis some POD-NS residuals (this is an iterative version of the first stabilization method). When the size of the basis becomes too large, a new POD compression is performed. This method gives very good results for the 1*D* Burgers equation, but convergence is too slow for the 2*D* Navier–Stokes equations, at least in our configuration. One explanation is that the "missing" scales (when dynamics evolves) are not necessary "fine" scales, and thus the approximation using POD-NS residuals is not good. The second method is a hybrid method that couples DNS and reduced order models. The idea is to modify and update the database when dynamical evolution occurs, so that the leading basis functions always represent the updated dynamics. This method, coupled with a fast method to actualize the POD basis functions using this new database, gives good results if a sufficient amount of DNS is performed. Approximatively 20% of the total numerical costs can be saved using such hybrid method.

#### References

- T. Bui-Thanh, K. Willcox, O. Ghattas, B. van Bloemen Waander, Goal-oriented, model-constrained optimization for reduction of large-scale systems, J. Comp. Phys. 224 (2) (2007) 880–896.
- [2] P. Holmes, J.L. Lumley, G. Berkooz, Turbulence, Coherent Structures, Dynamical Systems and Symmetry, Cambridge Monographs on Mechanics, 1996. [3] L. Sirovich, Turbulence and the dynamics of coherent structures, Quart. Appl. Math. XLV (3) (1987) 561–590.
- [4] L. Cordier, M. Bergmann, Proper orthogonal decomposition: an overview, in: Lecture Series 2002–04 on Post-processing of Experimental and Numerical Data, Von Kármán Institute for Fluid Dynamics, 2002.
- [5] J. Nocedal, S.J. Wright, Numerical Optimization, Springer Series in Operations Research, 1999.
- [6] C.W. Rowley, Model reduction for fluids, using balanced proper orthogonal decomposition, Int. J. Bifurc. Chaos 15 (3) (2005) 997-1013.
- [7] K. Willcox, J. Peraire, Balanced model reduction via the proper orthogonal decomposition, AIAA J. 40 (11) (2002) 2323–2330.

 [8] E. Akervik, G. H
 *a*pffner, U. Ehrenstein, D.S. Henningson, Optimal growth, model reduction and control in a separated boundary-layer flow using global eigenmodes, J. Fluid Mech. 579 (2007) 305–314.

- [9] B.R. Noack, H. Eckelman, A low dimensional Galerkin method for the three-dimensional flow around a circular cylinder, Phys. Fluids 6 (1) (1994) 124– 143.
- [10] D. Rempfer, On low-dimensional Galerkin models for fluid flow, Theor. Comput. Fluid Dyn. 14 (2000) 75-88.
- [11] B.R. Noack, P. Papas, P.A. Monkewitz, The need for a pressure-term representation in empirical Galerkin models of incompressible shear-flows, J. Fluid Mech. 523 (2005) 339-365.
- [12] B. Galletti, C.-H. Bruneau, L. Zannetti, A. Iollo, Low-order modelling of laminar flow regimes past a confined square cylinder, J. Fluid Mech. 503 (2004) 161–170.
- [13] P. Sagaut, Large-eddy Simulation for Incompressible Flows An Introduction, Springer-Verlag, 2005.
- [14] Y. Bazilevs, V.M. Calo, J.A. Cottrell, T.J.R. Hugues, A. Reali, G. Scovazzi, Variational multiscale residual-based turbulence modeling for large eddy simulation of incompressible flows, Comput. Meth. Appl. Mech. Eng. 197 (2007) 173–201.
- [15] R.D. Prabhu, S.S. Collis, Y. Chang, The influence of control on proper orthogonal decomposition of wall-bounded turbulent flows, Phys. Fluids 13 (2) (2001) 520–537.
- [16] B.R. Noack, K. Afanasiev, M. Morzyński, G. Tadmor, F. Thiele, A hierarchy of low-dimensional models for the transient and post-transient cylinder wake, J. Fluid Mech. 497 (2003) 335–363.
- [17] M. Bergmann, L. Cordier, Optimal control of the cylinder wake in the laminar regime by trust-region methods and pod reduced-order models, J. Comp. Phys. 227 (16) (2008) 7813–7840.
- [18] J.L. Lumley, Atmospheric turbulence and wave propagation, in: A.M. Yaglom, V.I. Tatarski, The Structure of Inhomogeneous Turbulence, 1967, pp. 166– 178.
- [19] L. Cordier, M. Bergmann, Two typical applications of POD: coherent structures eduction and reduced order modelling, in: Lecture Series 2002–04 on Post-processing of Experimental and Numerical Data, Von Kármán Institute for Fluid Dynamics, 2002.
- [20] S. Sirisup, G.E. Karniadakis, A spectral viscosity method for correcting the long-term behavior of POD model, J. Comp. Phys. 194 (2004) 92–116.
- [21] A.E. Deane, I.G. Kevrekidis, G.E. Karniadakis, S.A. Orszag, Low-dimensional models for complex geometry flows: application to grooved channels and circular cylinders, Phys. Fluids 3 (10) (1991) 2337–2354.
- [22] M. Bergmann, Optimisation aérodynamique par réduction de modéle POD et contrôle optimal, Application au sillage laminaire d'un cylindre circulaire, Ph.D. Thesis, Institut National Polytechnique de Lorraine, Nancy, France, 2004.
- [23] N. Aubry, P. Holmes, J.L. Lumley, E. Stone, The dynamics of coherent structures in the wall region of a turbulent boundary layer, J. Fluid Mech. 192 (1988) 115–173.
- [24] B. Podvin, J. Lumley, A low-dimensional approach for the minimal flow unit, J. Fluid Mech. 362 (1998) 121-151.
- [25] J. Delville, L. Ukeiley, L. Cordier, J.-P. Bonnet, M. Glauser, Examination of large-scale structures in a turbulent mixing layer. Part 1. Proper orthogonal decomposition, J. Fluid Mech. 391 (1999) 91–122.
- [26] D. Rempfer, H.F. Fasel, Evolution of three-dimensional coherent structures in a flat-plate boundary layer, J. Fluid Mech. 260 (1994) 351–375.
- [27] D. Rempfer, Investigations of boundary layer transition via Galerkin Projections on Empirical Eigenfunctions, Phys. Fluids 8 (1) (1996) 175-188.
- [28] G.S. Karamanos, G.E. Karniadakis, A spectral vanishing viscosity method for large eddy simulations, J. Comp. Phys. 162 (2000) 22-50.
- [29] M. Bergmann, L. Cordier, J.-P. Brancher, Optimal rotary control of the cylinder wake using POD reduced order model, Phys. Fluids 17 (9) (2005) 097101:1-097101:21.
- [30] M. Buffoni, S. Camarri, A. Iollo, M. Salvetti, Low-dimensional modelling of a confined three-dimensional wake flow, J. Fluid Mech. 569 (2006) 141-150.
- [31] M. Couplet, C. Basdevant, P. Sagaut, Calibrated reduced-order POD–Galerkin system for fluid flow modelling, J. Comp. Phys. 207 (2005) 192–220.
- [32] V.L. Kalb, A.E. Deane, An intrinsic stabilization scheme for proper orthogonal decomposition based low-dimensional models, Phys. Fluids 19 (2007) 054106.
- [33] W. Cazemier, Proper orthogonal decomposition and low-dimensional models for turbulent flows, Ph.D. Thesis, université de Groningen 1997.
- [34] W. Cazemier, R.W.C.P. Verstappen, A.E.P. Veldman, Proper orthogonal decomposition and low-dimensional models for driven cavity flows, Phys. Fluids 10 (7) (1998) 1685–1699.
- [35] B.R. Noack, M. Schlegel, B. Ahlborn, G. Mutschke, M. Morzynski, P. Comte, G. Tadmor, A finite-time thermodynamics formalism for unsteady flows, J. Non-Equilib. Thermodyn. 3 (2) (2008) 103–148.
- [36] G. Vigo, The Proper Orthogonal Decomposition applied to unsteady compressible Navier–Stokes equation, Tech. Rep. 3945, INRIA, 1998.
- [37] A. Iollo, A. Dervieux, J.A. Désiéri, S. Lanteri, Two stable pod-based approximations to the Navier-Stokes equations, Comput. Visual. Sci. 3 (2000) 61-66.
- [38] A. Iollo, S. Lanteri, J.A. Désiéri, Stability properties of POD-Galerkin approximations for the compressible Navier-Stokes equations, Theoret. Comput. Fluid Dyn. 13 (2000) 377-393.
- [39] T.J.R. Hugues, G. Feijóo, L. Mazzei, J.B. Quincy, The variational multiscale method—a paradigm for computational mechanics, Comput. Meth. Appl. Mech. Eng. 166 (1998) 3–24.
- [40] K. Ito, S.S. Ravindran, A reduced-order method for simulation and control of fluid flows, J. Comp. Phys. 143 (1998) 403-425.
- [41] S.S. Ravindran, A reduced-order approach for optimal control of fluids using proper orthogonal decomposition, Int. J. Numer. Meth. Fluids 34 (2000) 425–448.
- [42] W.R. Graham, J. Peraire, K.T. Tang, Optimal control of vortex shedding using low order models. Part 2: Model-based control, Int. J. Numer. Meth. Eng. 44 (7) (1999) 973–990.
- [43] X. Ma, G.E. Karniadakis, A low-dimensional model for simulating three-dimensional cylinder flow, J. Fluid Mech. 458 (2002) 181–190.
- [44] J. Burkardt, M.D. Gunzburger, H.-C. Lee, Centroidal Voronoi tessellation-based reduced-order modeling of complex systems, Tech. Rep., Florida State University, 2004.
- [45] W.R. Graham, J. Peraire, K.T. Tang, Optimal control of vortex shedding using low order models. Part 1. Open-loop model development, Int. J. Numer. Meth. Eng. 44 (7) (1999) 945–972.
- [46] M. Fahl, Trust-region methods for flow control based on reduced order modeling, Ph.D. Thesis, Trier University, 2000.
- [47] E. Arian, M. Fahl, E.W. Sachs, Trust-region proper orthogonal decomposition for flow control, Icase Report 2000–25.
- [48] Y. Saad, M.H. Schultz, Gmres: a generalized residual algorithm for solving nonsymmetric linear systems, SIAM J. Sci. Stat. Comput. 7 (3) (1986) 856-869.
- [49] S. Sirisup, G.E. Karniadakis, D. Xiu, I.G. Kevrekidis, Equation-free/Galerkin-free pod-assisted computation of incompressible flows, J. Comp. Phys. 207 (2005) 568–587.